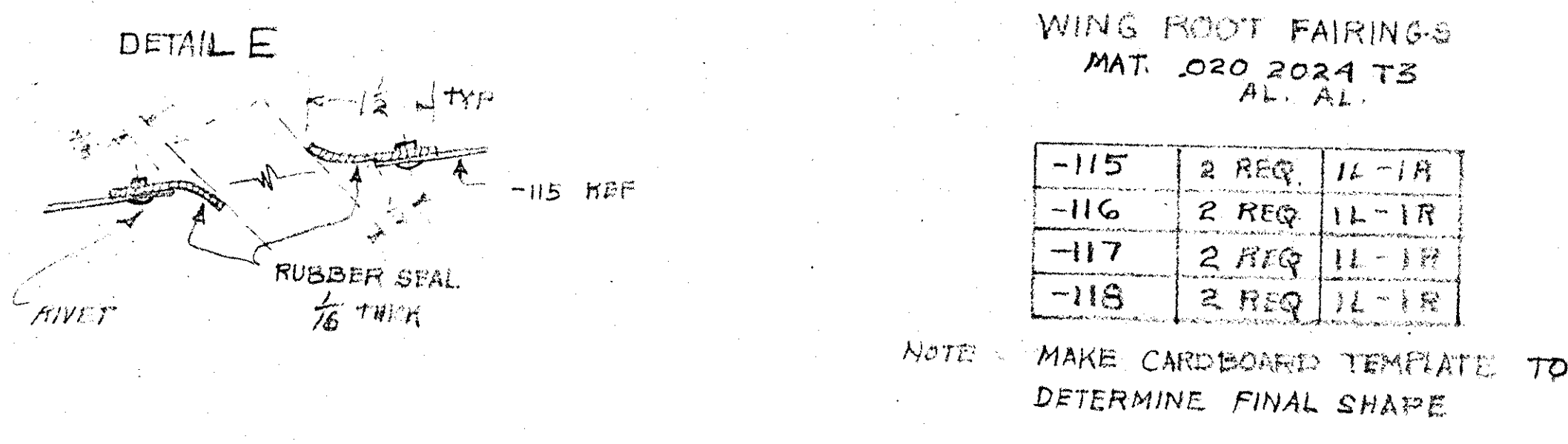
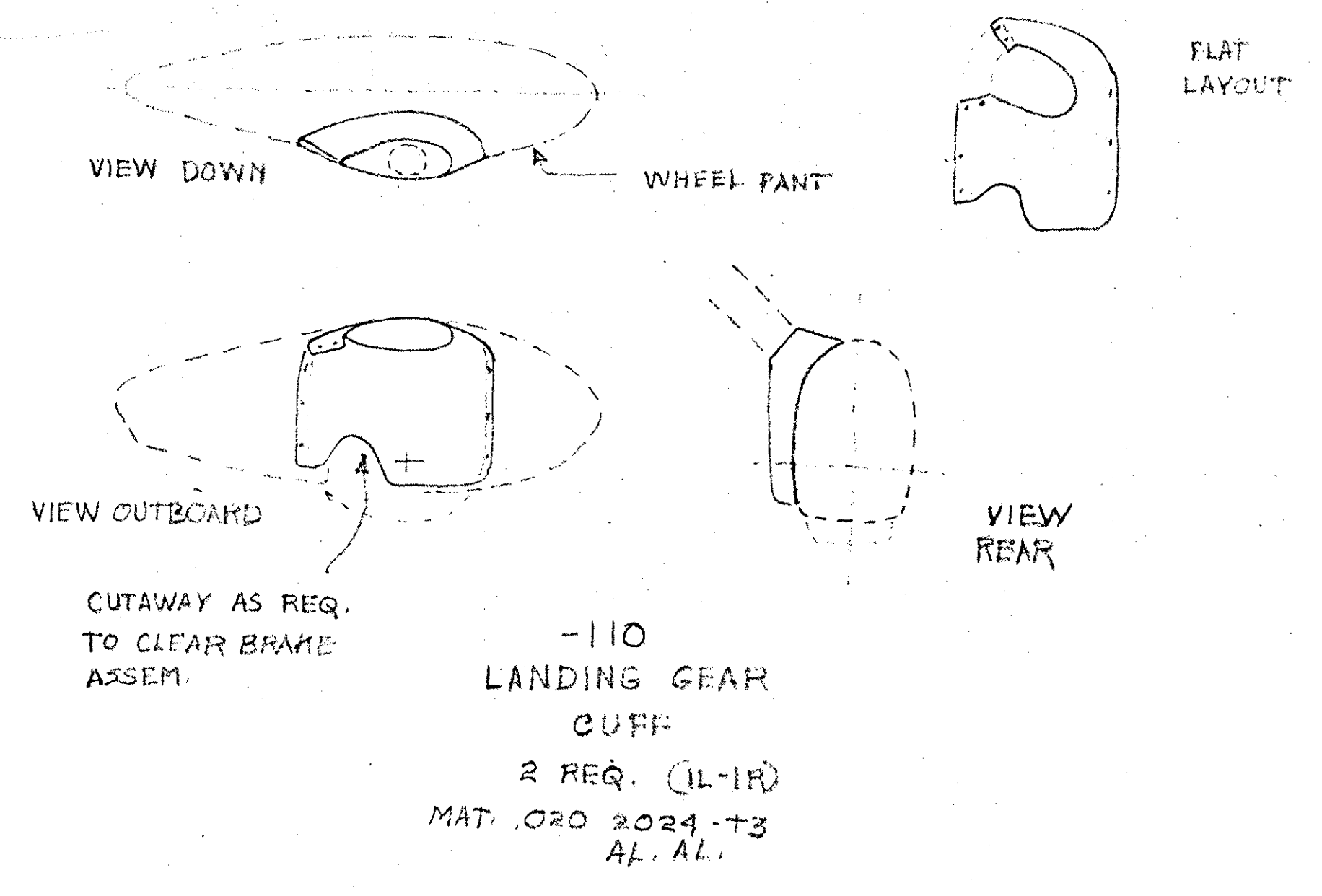


- NOTES:
1. FOR COMPLETE WEIGHT AND BALANCE PROCEDURE, CONSULT FAA, PART AC 15.13-1
 2. WEIGHT POINTS = 2 MAIN WHEELS AND TAIL WHEEL
 3. DATUM = FORWARD FACE OF FIREWALL
 4. LEVELING POINTS = CROSS TUBE AT FRONT INSTRUMENT PANEL AND LONGERON IN FRONT COCKPIT
 5. ALL MEASUREMENTS ARE TAKEN FROM DATUM
 6. WEIGH WITHOUT OIL OR FUEL
 7. FOR OIL CAP = USE ENGINE MFG. SPEC. FOR LOCATION = USE ACTUAL MEASUREMENT
 8. DATUM TO MAIN WHEEL = USE ACTUAL MEASUREMENT
 9. WHEEL BASE = USE ACTUAL MEASUREMENT
 10. CENTER OF GRAVITY RANGE = +30 IN TO +34 IN AFT OF DATUM
 11. OIL WEIGHT = 7 1/2 LB. PER GAL. (US)
FUEL WT = 6 LB. PER GAL. (US)
PASS. WT. = 170 LBS.
PILOT WT = 170 LBS.



-115	2 REQ	1L-1R
-116	2 REQ	1L-1R
-117	2 REQ	1L-1R
-118	2 REQ	1L-1R

MARQUART MA-5		ED MARQUART PO. BOX 3052 RIVERSIDE CAL. 92519	
"Charger"			
DRN	71	FAIRINGS - COVERS	
CHK	76	WT. AND BALANCE	
REV.	76		
TR.	78		
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